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B.Tech. 7th Semester End-Term Examination

ME

APPLIED THERMODYNAMICS - II

(New Regulation w.e.f. 2017-18 & New Syllabus w.e.f. 2018-19)

Full Marks – 70

Time – Three hours

The figures in the margin indicate full marks for the questions.

1. Answer any four of the following :

(4 × 8 = 32)

- 'There are certain disadvantages of increasing the delivery pressure to a high value in reciprocating air compressors' – elaborate the statement with the help of suitable diagrams.
- Obtain a mathematical relationship for optimum pressure ratio in a gas turbine plant for fixed values of inlet temperatures of the compressor and the turbine of the plant.
- The criterion for the thermodynamic efficiency of a reciprocating compressor is isothermal compression, while for rotary compressor it is isentropic compression. Discuss the reasons for this.
- Design a jet engine that alternately accelerates a contained mass of air rearward and then breathes in a fresh mass of air to replace it. The engine can be made with few or no moving parts.
- A refrigerant is superheated at the end of compression and under-cooled before throttling in a vapour compression refrigeration cycle. Represent the cycle of operation in T-s and p-h diagrams. Also find the theoretical COP of the cycle from the diagrams.
- Processes applied for summer and winter air-conditioning are different – elaborate the statement with suitable diagrams.

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2. Answer any two of the following:

(2 × 9 = 18)

- (a) The following observations are recorded during a trial on a two-stage single-acting reciprocating air compressor. The free air delivered is 6 m³/min. Atmospheric pressure and temperature are 1 bar and 27°C respectively. Delivery pressure is 40 bar. Speed of compressor is 400 rpm. The intermediate pressure is 6 bar. Temperature at the inlet to the second stage is 27°C. The law of compression is $pv^{1.3} = C$. The mechanical efficiency is 80%. Stroke of LP compressor = Diameter of LP Compressor = Stroke of HP compressor. Find the power required and diameters of the cylinders.
- (b) In an ideal jet propulsion cycle, air enters the compressor at 1 atm, 15°C. The pressure of air leaving the compressor is 5 atm and the maximum temperature is 870°C. The air expands in the turbine to such a pressure that the turbine work is equal to the compressor work. On leaving the turbine, the air expands reversibly and adiabatically in a nozzle to 1 atm. Determine the velocity of air leaving the nozzle.
- (c) A gas turbine unit receives air at 1 bar and 300 K and compresses it adiabatically to 6.2 bar. The compressor efficiency is 85%. The fuel has a heating value of 44 186 kJ/kg and fuel-air ratio is 0.02 kg of fuel per kg of air. The turbine internal efficiency is 90%. Calculate the work of turbine and compressor per kg of air and thermal efficiency of the cycle. For products of combustion, $C_p = 1.147$ kJ/kgK and $\gamma = 1.333$.

3. Answer any four of the following:

(4 × 5 = 20)

- (a) For a single-stage reciprocating compressor with pressure ratio 4, find the maximum clearance at which the volumetric efficiency becomes zero. Take $n=1.3$.
- (b) For efficient run of a compressor, surging zone in the head-capacity characteristics is avoided – Elaborate.
- (c) Draw the velocity vector diagrams for the impeller of a centrifugal compressor and show how the outlet velocity diagram is affected by slip.
- (d) In a vapour compression refrigeration plant, the enthalpy values at different points are: enthalpy at the exit of the evaporator = 185 kJ/kg, enthalpy at exit of the compressor = 210 kJ/kg, enthalpy at exit of the condenser = 85 kJ/kg. What is the power required per kW of cooling to be produced?
- (e) Calculate the amount of heat removed per kg of dry air if the initial condition of air is 35°C, 70% RH and the final condition is 25°C, 60% RH.
- (f) R – 134a is an eco-friendly refrigerant – write a note on the comment comparing it with other refrigerants.

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